

## STRUCTURAL DYNAMIC COMPUTING

### Reduction methods for dynamic systems:

- Static condensation
- Modal truncation
- Generalized SDOF systems
- Use of Ritz vectors

## REDUCTION OF DOF'S - OVERVIEW

Computational savings can be done by the following methods:

### \* static condensation

Dof's without inertia are eliminated

### \* Modal truncation

Higher modes are neglected in the modal expansion

### \* Generalized SDOF system

An approximation of the first mode is used to obtain a SDOF system

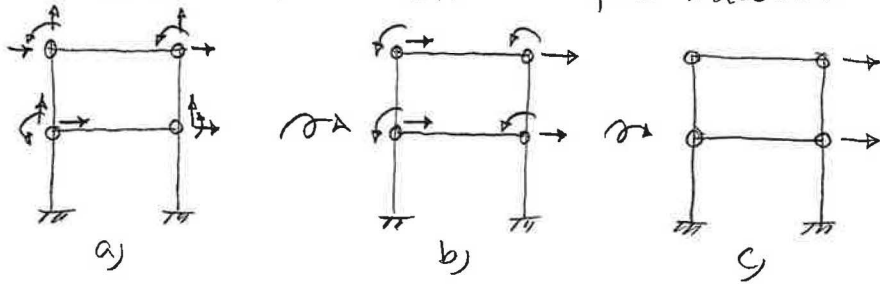
### \* Rayleigh-Ritz method

A special set of base vectors are used to express the displacements

# STATIC CONDENSATION OF DOFS WITHOUT INERTIA

①

consider a frame with lumped masses:



a) All degrees of freedom.

b) Horizontal excitation  $\Rightarrow$  vertical dofs not needed in dynamic analysis

c) Beams are stiff in axial direction and zero diagonal elements in rotational dofs

starting from a) dofs with zero inertia have displacements  $u_0$

The remaining (two) dofs are dynamic dofs  $u_c$ .

Partitioned system:

$$\begin{bmatrix} m_{cc} & 0 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} \ddot{u}_c \\ \ddot{u}_0 \end{bmatrix} + \begin{bmatrix} k_{cc} & k_{co} \\ k_{co}^T & k_{oo} \end{bmatrix} \begin{bmatrix} u_c \\ u_0 \end{bmatrix} = \begin{bmatrix} p_c \\ 0 \end{bmatrix} \dots (1)$$

STATIC COND. - cont.

②

Eq. (1b) can be used to eliminate  $u_0$  in eq. (1a):

$$k_{co}^T u_c + k_{oo} u_0 = 0 \quad ; \quad u_0 = -k_{oo}^{-1} k_{co}^T u_c \dots (2)$$

Insert (2) into (1)  $\Rightarrow$

$$m_{cc} \ddot{u}_c + \underbrace{(k_{cc} + k_{co} (-k_{oo}^{-1} k_{co}^T))}_{\hat{k}_{cc}} u_c = p_c(t)$$

The condensed stiffness matrix:

$$\hat{k}_{cc} = k_{cc} - k_{co} k_{oo}^{-1} k_{co}^T$$

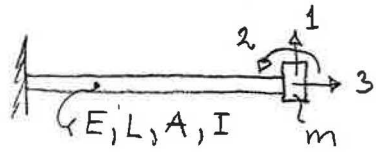
The condensed system:

$$m_{cc} \ddot{u}_c + \hat{k}_{cc} u_c = p_c(t)$$

determines the dynamic dofs

Displacements in remaining dofs are found from (2).

# EXAMPLE : STATIC CONDENSATION

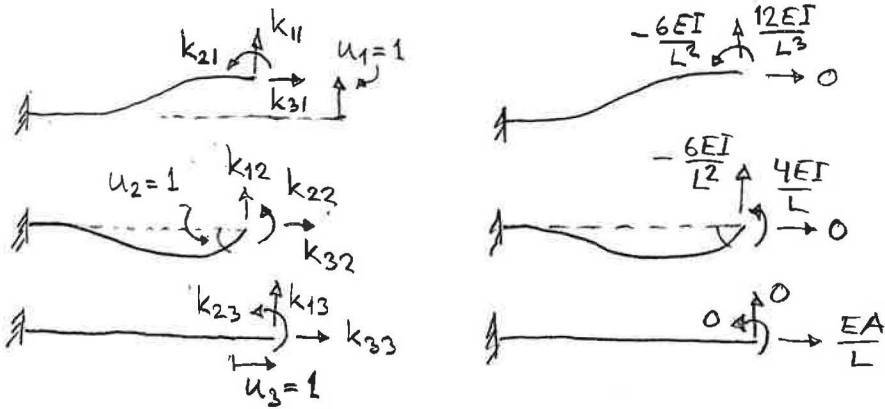


Cantilever beam:  
Eliminate dofs 2 and 3.

Inertia in dof 1 only\*

$$m = \begin{bmatrix} m & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix} \quad k = \begin{bmatrix} k_{11} & k_{12} & k_{13} \\ k_{21} & k_{22} & k_{23} \\ k_{31} & k_{32} & k_{33} \end{bmatrix}$$

Establish the stiffness matrix :



Giving :  $k = \begin{bmatrix} \frac{12EI}{L^3} & -\frac{6EI}{L^2} & 0 \\ -\frac{6EI}{L^2} & \frac{4EI}{L} & 0 \\ 0 & 0 & \frac{EA}{L} \end{bmatrix}$

Rem. \*) Moment of inertia small and longitudinal stiffness is high. #

# EXAMPLE CONT.

Partitioning :

$$\begin{bmatrix} m_{tt} & 0 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} \ddot{u}_{tt} \\ u_o \end{bmatrix} + \begin{bmatrix} k_{tt} & k_{to} \\ k_{ot} & k_{oo} \end{bmatrix} \begin{bmatrix} u_t \\ u_o \end{bmatrix} = \begin{bmatrix} P_t(t) \\ 0 \end{bmatrix}$$

Eq. of motion :

$$m_{tt} \ddot{u}_t + \hat{k}_{tt} u_t = P_t(t) ;$$

$$\hat{k}_{tt} = k_{tt} - k_{ot}^T k_{oo}^{-1} k_{ot} =$$

$$= \frac{12EI}{L^3} - \begin{bmatrix} -\frac{6EI}{L^2} & 0 \end{bmatrix} \begin{bmatrix} \frac{L}{4EI} & 0 \\ 0 & \frac{L}{EA} \end{bmatrix} \begin{bmatrix} -\frac{6EI}{L^2} \\ 0 \end{bmatrix} =$$

$$= \frac{12EI}{L^3} - \begin{bmatrix} -\frac{6EI}{L^2} & 0 \end{bmatrix} \begin{bmatrix} -\frac{6EI}{L^2} \\ 0 \end{bmatrix} =$$

$$= \frac{12EI}{L^3} - \frac{36EI}{4L^3} = \frac{12EI}{L^3} - \frac{9EI}{L^3} = \frac{3EI}{L^3}$$

Compare with cantilever 1D :



The same eq.

$$m \ddot{u} + \frac{3EI}{L^3} u = p(t)$$

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# TRUNCATED MODAL EXPANSION

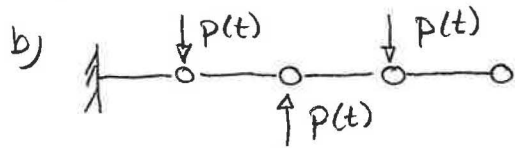
The modal expansion is truncated at  $r < N$  giving an approximation:

$$u \approx \sum_1^r \phi_n q_n$$

Higher modes are left out.

Modal coordinates are obtained for  $r$  uncoupled equations.

The approximation is good if the spatial distribution of external forces triggers the lower modes:

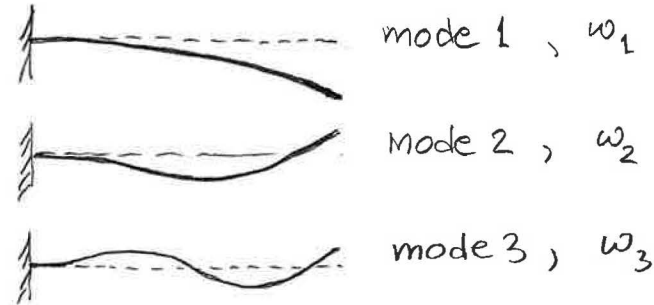


$u$  is well described by the lower modes in a)

Higher modes have bigger influence in b)

# TRUNCATED cont.

However, the time variation of the force is also important. Consider case a) again for increasing frequency of the load



Despite the spatial distribution of the load, higher modes are triggered as the frequency increases;

$$\left\{ \begin{array}{l} \omega \approx \omega_1 \Rightarrow \text{mode 1 behavior} \\ \omega_1 < \omega \approx \omega_2 \Rightarrow \text{mode 2} \text{ ---''---} \\ \omega_2 < \omega \approx \omega_3 \Rightarrow \text{---''--- 3 ---''---} \end{array} \right.$$

Conclusion:

The modes that are triggered depend both on the spatial distribution and on the time variation of the loading.

## OBTAINING THE REDUCED SYSTEM BY MODAL REDUCTION

After having chosen the number  $r$  of the modes to keep, the modal expansion gives an approximation of the displacement vector:

$$u \approx \sum_1^r \Phi_n q_n = \Phi_r q_r \dots (*)$$

The original system  $\xrightarrow{[N \times r]} [r \times 1]$

$$m \ddot{u} + k u = p$$

is diagonalized by a transformation similar to the case of the full system:

$$\underbrace{\Phi_r^T m \Phi_r}_{M_r} \ddot{q}_r + \underbrace{\Phi_r^T k \Phi_r}_{K_r} q_r = \underbrace{\Phi_r^T p}_{P_r} \dots (**)$$

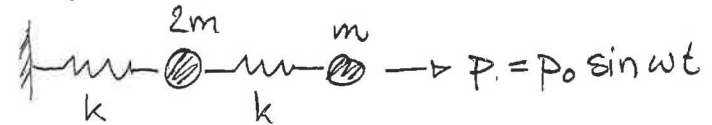
Dimensions of matrices

$$\left\{ \begin{array}{l} M_r \text{ and } K_r : [r \times r][r \times r] = [r \times r] \\ q_r \text{ and } P_r : [r \times 1] \end{array} \right.$$

The system (\*\*) is solved as decoupled SDOF equations giving  $q_r$ . Then the physical displacement  $u$  are found from (\*).

## EX. MODAL TRUNCATION - TWO TO ONE DOF

Look again at the two dof system being diagonalized and solved for harmonic loading previously



$$k = \begin{bmatrix} 2 & -1 \\ -1 & 1 \end{bmatrix} \quad m = \begin{bmatrix} 2 & 0 \\ 0 & 1 \end{bmatrix} \quad P = p \begin{bmatrix} 0 \\ 1 \end{bmatrix}$$

It was given that

$$\left\{ \begin{array}{l} \omega_1^2 = (1 - \frac{1}{\sqrt{2}}) \frac{k}{m}, \quad \Phi_1 = \begin{bmatrix} \frac{1}{\sqrt{2}} \\ 1 \end{bmatrix} \\ \omega_2^2 = (1 + \frac{1}{\sqrt{2}}) \frac{k}{m}, \quad \Phi_2 = \begin{bmatrix} \frac{1}{\sqrt{2}} \\ -1 \end{bmatrix} \end{array} \right.$$

Modal expansion

$$u = \Phi q = \begin{bmatrix} \frac{1}{\sqrt{2}} & \frac{1}{\sqrt{2}} \\ 1 & -1 \end{bmatrix} \begin{bmatrix} q_1 \\ q_2 \end{bmatrix}$$

Truncate according to

$$u \approx \Phi_r q_r = \begin{bmatrix} \frac{1}{\sqrt{2}} \\ 1 \end{bmatrix} q \dots (*) \quad (\text{one dof; } q)$$

obtain the reduced system:  $M_r \ddot{q}_r + K_r q_r = P_r$

$$M_r = \Phi_r^T m \Phi_r, \quad K_r = \Phi_r^T k \Phi_r \quad \text{and} \quad P_r = \Phi_r^T p$$

Ex modal trunc. cont.

$$M_z = \begin{bmatrix} \frac{1}{\sqrt{2}} & 1 \end{bmatrix} \begin{bmatrix} 2 & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} \frac{1}{\sqrt{2}} \\ 1 \end{bmatrix} m = \dots = 2m$$

$$K_z = \begin{bmatrix} \frac{1}{\sqrt{2}} & 1 \end{bmatrix} \begin{bmatrix} 2 & -1 \\ -1 & 1 \end{bmatrix} \begin{bmatrix} \frac{1}{\sqrt{2}} \\ 1 \end{bmatrix} k = \dots = (2 - \sqrt{2})k$$

$$F_z = \begin{bmatrix} \frac{1}{\sqrt{2}} & 1 \end{bmatrix} \begin{bmatrix} 0 \\ 1 \end{bmatrix} P = P$$

The reduced system

$$2m \ddot{q} + (2 - \sqrt{2})k q = P ;$$

$$\ddot{q} + \underbrace{\left(1 - \frac{1}{\sqrt{2}}\right) \frac{k}{m}}_{\omega_1^2} q = \frac{P}{2m}$$

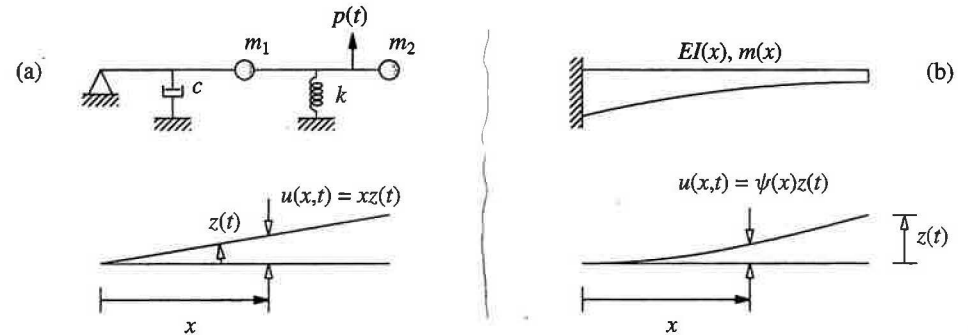
It is clear that the reduced system behaves according to the first mode

This is also seen from (\*):

$$\begin{bmatrix} u_1 \\ u_2 \end{bmatrix} = \begin{bmatrix} \frac{q}{\sqrt{2}} \\ q \end{bmatrix} \quad \text{the masses are in phase according mode 1}$$

## GENERALIZED SDOF SYSTEMS

systems with more than one discrete mass or even distributed mass can be treated as a SDOF system by assuming a given deflected shape  $\psi(x)$



The generalized equation of motion is written as

$$\tilde{m} \ddot{z} + \tilde{c} \dot{z} + \tilde{k} z = \tilde{p}(t)$$

The generalized quantities are determined from  $\psi(x)$

Useful for determining the lowest natural frequency

$$\omega_n^2 = \frac{\tilde{k}}{\tilde{m}}$$

## GENERALIZED SDOF SYSTEMS

Knowledge about the system behaviour can be used to form a SDOF system from a MDOF system:

$$m \ddot{u} + k u = p(t) \quad \dots (*)$$

Assume the displacements follow a particular shape  $\Psi$  ( $N \times 1$ )

$$u = \Psi \cdot z(t) \quad \dots (**)$$

controlled by a single coordinate  $z(t)$ .

$$\Rightarrow \ddot{u} = \Psi \ddot{z}(t) \quad \text{insert into } (*) \Rightarrow$$

$$m \Psi \ddot{z} + k \Psi z = p$$

Multiply by  $\Psi^T \Rightarrow$

$$\underbrace{\Psi^T m \Psi}_{m^*} \ddot{z} + \underbrace{\Psi^T k \Psi}_{k^*} z = \underbrace{\Psi^T p}_{p^*}$$

The generalized SDOF-system:

$$\underline{m^* \ddot{z} + k^* z = p^*}$$

The MDOF system has been reduced to a SDOF system by the assumption of a deflected shape  $\Psi$  and a single coordinate  $z(t)$  controlling the motion.

GEN. SDOF CONT.

The generalized systems initial value problem can be solved by using (\*\*)

$$u = \Psi \cdot z ; \quad \Psi^T u = \Psi^T \Psi z ;$$
$$z = \frac{\Psi^T u}{\Psi^T \Psi} \quad (\Psi^T \Psi \text{ is a scalar})$$

The initial values for the generalized system are thus given by

$$\begin{cases} z(0) = \frac{\Psi^T u(0)}{\Psi^T \Psi} \\ \dot{z}(0) = \frac{\Psi^T \dot{u}(0)}{\Psi^T \Psi} \end{cases}$$

with  $u(0)$  and  $\dot{u}(0)$  being physical initial value vectors of the original mdof system.

## ESTIMATING THE LOWEST NATURAL FREQUENCY

Natural frequencies are determined from the homogeneous system:

$$(k - \omega^2 m) \phi = 0$$

Assume an approximation of the first mode given by  $\phi_1 \approx \Psi \Rightarrow$

$$(k - \omega_1^2 m) \Psi \approx 0 ; \quad k \Psi \approx \omega_1^2 m \Psi ;$$

Multiply by  $\Psi^T \Rightarrow \Psi^T k \Psi \approx \omega_1^2 \Psi^T m \Psi ;$

$$\omega_1^2 \approx \frac{\Psi^T k \Psi}{\Psi^T m \Psi}$$

This is the so called Rayleigh quotient

Rem. Note the resemblance with strain energy  $E_s$  and kinetic energy  $E_k$

$$E_s = \frac{1}{2} u^T k u \quad \text{and} \quad E_k = \frac{1}{2} \dot{u}^T m \dot{u}$$

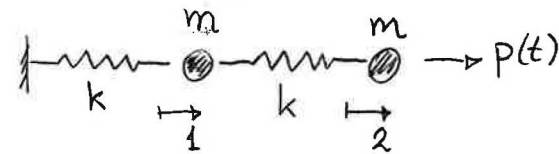
$$\dot{u} = \omega u \quad \text{and} \quad E_s = E_k \quad \Rightarrow$$

$$\frac{1}{2} u^T k u = \omega^2 \frac{1}{2} u^T m u ; \quad \omega^2 = \frac{u^T k u}{u^T m u}$$

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## Ex. GENERALIZED SDOF SYSTEM

Consider again the simple two dof system



$$k = k \begin{bmatrix} 2 & -1 \\ -1 & 1 \end{bmatrix} \quad m = m \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} \quad P = p(t) \begin{bmatrix} 0 \\ 1 \end{bmatrix}$$

Assume that the system moves in something like the first mode:

$$\Psi = \begin{bmatrix} 1 \\ 2 \end{bmatrix} \quad \text{a linear deflection}$$

$$u(t) = \Psi z(t) = \begin{bmatrix} z(t) \\ 2z(t) \end{bmatrix}$$

Generalized quantities:

$$\left. \begin{aligned} k^* &= [1 \ 2] k \begin{bmatrix} 2 & -1 \\ -1 & 1 \end{bmatrix} \begin{bmatrix} 1 \\ 2 \end{bmatrix} = [1 \ 2] k \begin{bmatrix} 0 \\ 1 \end{bmatrix} = 2k \\ m^* &= [1 \ 2] m \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} 1 \\ 2 \end{bmatrix} = m [1 \ 2] \begin{bmatrix} 1 \\ 2 \end{bmatrix} = 5m \\ P^* &= [1 \ 2] \begin{bmatrix} 0 \\ 1 \end{bmatrix} p(t) = 2p(t) \end{aligned} \right\}$$



Ex. cont.

The generalized SDOF system:

$$5m \ddot{z}(t) + 2kz(t) = 2p(t)$$

The natural frequency  $\omega^2 = \frac{2k}{5m} = 0.4 \frac{k}{m}$

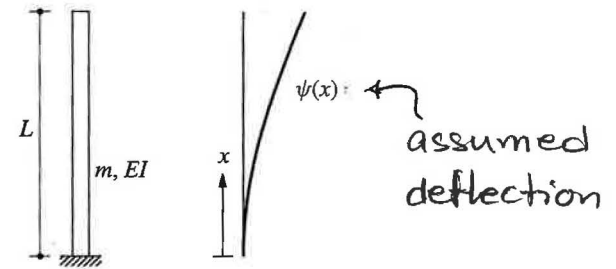
Compare with the lowest natural frequency:

$$\omega_1^2 = 0.382 \frac{k}{m}$$

The estimated frequency is a good approximation of the lowest natural frequency

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## BEAM-LIKE STRUCTURES - GENERALIZED SDOF SYSTEMS



Generalized quantities

$$\begin{cases} \tilde{m} = \int_0^L m(x) [\psi(x)]^2 dx \\ \tilde{k} = \int_0^L EI(x) [\psi''(x)]^2 dx \\ \tilde{p} = \int_0^L p(x,t) \psi(x) dx \end{cases}$$

in the equation

$$\tilde{m} \ddot{z} + \tilde{k} z = \tilde{p}(t)$$

for motion according to the restrictions given by  $\psi(x)$ .

Rem. Beam properties  $m$  and  $EI$  can vary with  $x$ .

Ex. Lowest natural frequency estimates for a cantilever

An estimate of the lowest natural frequency can be found from an assumed reasonable shape  $\psi(x)$  for the lowest vibration mode.

The assumed shape must satisfy geometric boundary conditions.

TABLE 8.5.1 NATURAL FREQUENCY ESTIMATES FOR A UNIFORM CANTILEVER

|    | $\psi(x)$              | $\alpha_n$ | % Error |
|----|------------------------|------------|---------|
| 1. | $3x^2/2L^2 - x^3/2L^3$ | 3.57       | 1.5     |
| 2. | $1 - \cos(\pi x/2L)$   | 3.66       | 4       |
| 3. | $x^2/L^2$              | 4.47       | 27      |

$$\omega_n = \alpha_n \sqrt{\frac{EI}{mL^4}}$$

Cantilever on previous page  $\omega_n$  for 3 assumed shapes in the table

The static deflection from distributed forces 1. is a good choice of  $\psi(x)$ . Trigonometric functions 2. is also accurate.

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APPROXIMATION BY RITZ VECTORS - OVERVIEW

A reduced approximate dynamic system can be obtained by Ritz vectors. A careful choice of a small number  $r$  of vectors ( $r \ll N$ ) yields a system with  $r$  dofs, giving a good approximation to the original system with  $N$  dofs.

$$u \approx \sum_1^r z_j(t) \psi_j = \Psi z(t), \quad [\Psi] = N \times r$$

Substituting into the standard form:

$$m \Psi \ddot{z} + c \Psi \dot{z} + k \Psi z = p(t)$$

Pre multiplying with  $\Psi^T \Rightarrow$

$$\tilde{m} \ddot{z} + \tilde{c} \dot{z} + \tilde{k} z = \tilde{p}(t)$$

The reduced system with

$$\tilde{m} = \Psi^T m \Psi, \quad \tilde{c} = \Psi^T c \Psi, \quad \tilde{k} = \Psi^T k \Psi \quad \text{and} \\ \tilde{p} = \Psi^T p$$

has  $r$  dofs as  $[z] = [\tilde{p}] = r \times 1$  and

$$[m] = [c] = [k] = r \times r.$$

The usefulness of the reduced system depends entirely on good choices of Ritz vectors

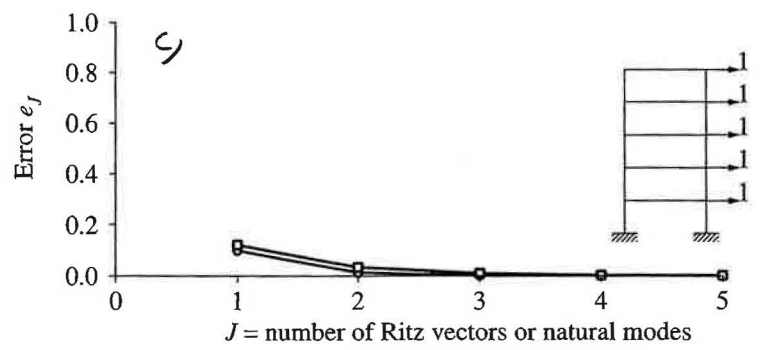
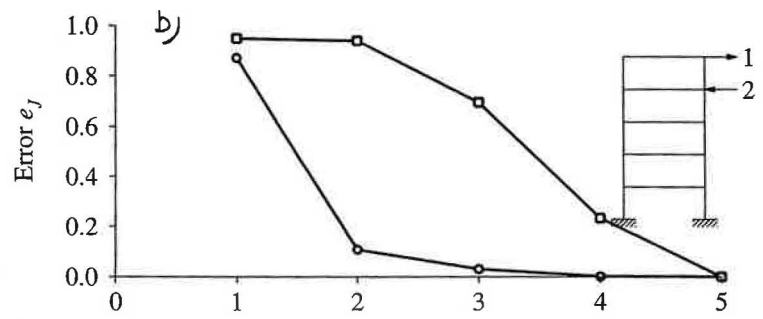
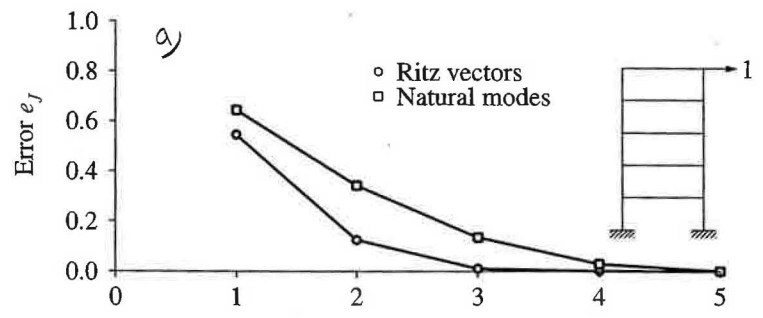
APPROXIMATION BY cont.

The choice of vectors can be based on:

- 1) Physical insight - good guess of "modes" that influence the system dynamics
- 2) An automatic procedure giving Ritz vectors that conforms with the spatial distribution of the load.

Rem. In 2) Gram-Schmidt's orthogonalization procedure is used in the so called Lanczos algorithm.

EXAMPLE : COMPARISON OF RITZ VECTORS AND NATURAL MODES WITH DIFFERENT FORCE DISTRIBUTIONS



case b) shows an distinct advantage of using base vectors conforming with the load distribution

APPLICATION RITZ VECTORS, IMPACT  
ON GLASS PANES

Two shape vectors were used:

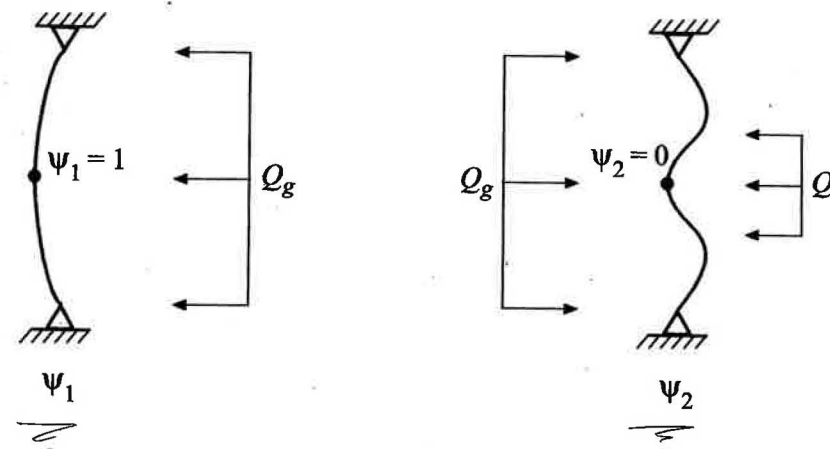


Figure 23: Construction of Ritz vectors.

**STRENGTH DESIGN METHODS  
FOR GLASS STRUCTURES**

MARIA FRÖLING

55

M. Fröling PhD-thesis

Structural  
Mechanics

Doctoral Thesis

## SOLVING THE REDUCED EQUATIONS - INITIAL VALUES

The reduced system ( $J$  dofs):

$$\tilde{m} \ddot{z} + \tilde{c} \dot{z} + \tilde{k} z = \tilde{p} \quad ; \quad u = \Psi z$$

How can initial values be invoked?

$$\begin{cases} u(0) = \Psi z(0) \\ \dot{u}(0) = \Psi \dot{z}(0) \end{cases} \quad [\Psi] = N \times J, \quad J \ll N$$

Multiply by  $\Psi^T \Rightarrow$

$$\begin{cases} \Psi^T u(0) = \Psi^T \Psi z(0) \\ \Psi^T \dot{u}(0) = \underbrace{\Psi^T \Psi}_{J \times J} \dot{z}(0) \end{cases}$$

We have two small linear systems of equations to solve in order to obtain initial values in  $z$ , i.e.

$z(0)$  and  $\dot{z}(0)$ .

## USING RITZ VECTORS TO OBTAIN APPROXIMATE NATURAL FREQ. & MODES

Linear combinations of Ritz vectors can also be used to approximate natural modes and frequencies by:

$$\tilde{\phi}_n = \Psi z_n \quad [z_n] = J \times 1$$

$\tilde{\phi}_n$  is the approximate eigenvector and  $z_n$  contains weight factors.

The best combination of Ritz vectors is found from the reduced eigen problem:

$$\hat{k} z = \rho \tilde{m} z$$

with the approximate natural circular frequencies:  $\omega_i^2 \approx \rho_i \quad i = 1, \dots, J$

USING RITZ VECT. CONT.

Derivations:

Eigenvector trial solution  $\tilde{\phi}$  as a linear combination of Ritz vectors

$$\tilde{\phi} = \Psi z \quad [N \times J] [J \times 1]$$

$$(k - \tilde{\omega}^2 m) \tilde{\phi} = 0 ;$$

$$\Psi^T (k - \tilde{\omega}^2 m) \Psi z = 0 ;$$

$$(\tilde{k} - \tilde{\omega}^2 \tilde{m}) z = 0 \quad \dots (*)$$

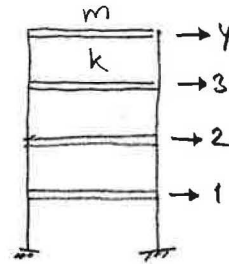
with  $\tilde{k} = \Psi^T k \Psi$  and  $\tilde{m} = \Psi^T m \Psi$   
 $\uparrow \quad \quad \quad \uparrow$   
 $\quad \quad \quad [J \times J]$

Good choice of  $\Psi \Rightarrow$

$$\tilde{\omega}_n \approx \omega_n \quad \text{for } n = 1, 2, \dots, J$$

The reduced system (\*) also yields the weights  $z_i$  giving the 'best' approximation of the true eigenvectors.

EX. FOUR STOREY FRAME - RITZ VECTORS

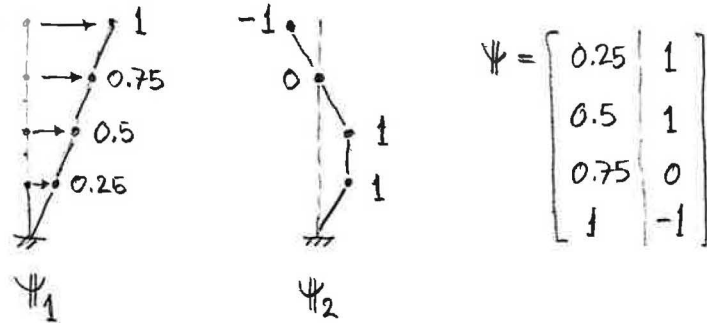


Approximate solution to eigenvalue problem:

$$(k - \omega^2 m) \phi = 0$$

$$m = m \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} ; \quad k = k \begin{bmatrix} 2 & -1 & 0 & 0 \\ -1 & 2 & -1 & 0 \\ 0 & -1 & 2 & -1 \\ 0 & 0 & -1 & 1 \end{bmatrix} \quad \text{Put } k=m=1$$

Add hoc choice of Ritz vectors:  $u = \Psi z$



Reduced system matrices:

$$\tilde{m} = \Psi^T m \Psi = \begin{bmatrix} 1.875 & -0.25 \\ -0.25 & 3.0 \end{bmatrix}$$

$$\tilde{k} = \Psi^T k \Psi = \begin{bmatrix} 0.25 & -0.25 \\ -0.25 & 3.0 \end{bmatrix}$$

EX FOUR STOREY, cont.

Solve the reduced eigen-value problem

$$(\tilde{k} - \tilde{\omega}^2 m) z = 0 \Rightarrow$$

$$\begin{cases} \tilde{\omega}_1^2 = 0.1236 & , & z_1 = \begin{bmatrix} -0.734 \\ -0.061 \end{bmatrix} \\ \omega_2^2 = 1.0 & , & z_2 = \begin{bmatrix} 0 \\ 0.577 \end{bmatrix} \end{cases}$$

"Best" approximations of  $\phi_1$  and  $\phi_2$  :

$$\Psi z_1 = \begin{bmatrix} -0.245 \\ -0.428 \\ -0.551 \\ -0.673 \end{bmatrix} \quad \Psi z_2 = \begin{bmatrix} 0.577 \\ 0.577 \\ 0 \\ -0.577 \end{bmatrix}$$

Exact solution:

$$\omega_1^2 = 0.1206$$

$$\omega_2^2 = 1$$

$$\phi_1 = \begin{bmatrix} -0.228 \\ -0.429 \\ -0.577 \\ -0.657 \end{bmatrix}$$

$$\phi_2 = \begin{bmatrix} 0.577 \\ 0.577 \\ 0 \\ -0.577 \end{bmatrix}$$

Exact!  $\Rightarrow$   
Good choice

An extension of the gen. SDOF method.

Orthogonality:

$$\phi_1^T \phi_2 = 0 \quad (\approx 10^{-17}), \quad \Psi_1^T \Psi_2 \neq 0 \quad (= -0.25)$$

$$\text{But } (\Psi z_1)^T \Psi z_2 = 0$$